

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

Listing of Claims:

1.(Currently Amended) Floor heating for an aircraft, ~~in particular for a freight aircraft~~, comprising a floor (20) made up of heatable panels (18), ~~characterised in that the panels (18) have running through them~~ defining a plurality of first hollow chambers (26) wherein each chamber has a first end, a second end, and is enclosed therebetween, and a feed line (28) operatively connected to the first ends of the first hollow chambers (26) for supplying thereto warm waste air which originates from the cooling of the aircraft's electronic equipment.

2.(Original) Floor heating in accordance with claim 1,
characterised in that the first hollow chambers (26) extend in the longitudinal direction of the aircraft inside the panels (18).

3.(Previously Presented) Floor heating in accordance with claim 1,
characterised in that the feed line (28) serves to connect the first hollow chambers (26) with an avionics bay (14) of the aircraft.

4.(Currently Amended) Floor heating in accordance with claim 1,
characterised in that the second ends of the first hollow chambers (26) are in flow connection with a plurality of second hollow chambers (32) indefined by the floor panels (34) of a an aft-located cargo hold door (24) of the aircraft.

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

5.(Original) Floor heating in accordance with claim 4,

characterised in that the second hollow chambers (32) terminate into the aircraft fuselage (10).

6.(Currently Amended) Floor heating in accordance with claim 1, ~~characterised in that~~further comprising:

a first bleed air ~~another feed line connects~~ operatively connecting the first ends of the first
hollow chambers (26) to a first supply of hot engine bleed air from the engine of the aircraft.

7.(Currently Amended) Floor heating in accordance with claim 6, wherein the second ends of the
first hollow chambers (26) are in flow connection with a plurality of second hollow chambers (32)
in defined by the flow panels (34) of an aft-located cargo hold door (24) of the aircraft, further
~~characterised in that~~comprising: a second bleed air feed line yet another feed line connects
operatively connecting the second hollow chambers (32) to a second supply of hot engine bleed
air from the engine of the aircraft.

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

8.(Currently Amended) Floor heating in accordance with claim ~~6~~7,

characterised in that the cross sections of the first and second bleed air feed lines ~~cross=~~
~~sections~~ determine the amount of hot engine bleed air supplied.

9.(Previously Presented) Floor heating in accordance with claim 1,

characterised in that the panels (18) are thermally uncoupled from a structure which
supports the floor (20).

10.(Previously Presented) Floor heating in accordance with claim 1,

characterised in that the panels (18) are provided with electric heating mats for
supplementary heating.

11.(Original) Floor heating in accordance with claim 10,

characterised in that the electric heating mats are positioned on the lower side of the
panels (18).

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

12.(Currently Amended) Floor heating in accordance with claim 1, ~~characterised in that~~ further comprising:

electric heating coils and/or wires ~~are~~ integrated into the first hollow chambers (26, ~~32~~)
for supplying supplementary heating therein.

13.(Currently Amended) Floor heating in accordance with claim 1, ~~characterised in that~~ further comprising:

ventilators ~~are~~ positioned in the first hollow chambers (26, ~~32~~) ~~in order~~ to generate a
forced flow through the first hollow chambers (26, ~~32~~).

14.(Previously Presented) Floor heating in accordance with claim 1,

characterised in that the panels (18) are provided with thermal insulation (42) on their
lower side.

15.(Previously Presented) Floor heating in accordance with claim 1,

characterised in that the panels (18) are profile elements produced by extrusion, in
particular by continuous extrusion.

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

16.(Currently Amended) Method for heating the floor of an aircraft, ~~in particular a freight aircraft, characterised in that the~~ comprising:

conveying warm waste air ~~originating from the cooling of electronic equipment in the aircraft is conveyed through a first plurality of hollow chambers in defined by the panels forming the floor, the warm waste air having originated from the cooling of electronic equipment of the~~ aircraft.

17.(Currently Amended) Method in accordance with claim 16,

characterised in that the warm waste air is conveyed through the panels in the longitudinal direction of the aircraft and ~~preferably~~ counter to the flight direction.

18.(Previously Presented) Method in accordance with claim 16,

characterised in that the warm waste air originates from the aircraft's avionics bay.

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

19.(Currently Amended) Method in accordance with claim 16,

characterised in that the warm waste air, after having flowed through the first plurality of
hollow chambers in the panels forming the floor, is thereafter conveyed through ~~the floor panels~~
~~of the aircraft's~~ that form a cargo hold door for the aircraft.

20.(Original) Method in accordance with claim 19,

characterised in that the warm waste air flows out into the aircraft fuselage after having
flowed through the floor panels of the cargo hold door.

21.(Currently Amended) Method in accordance with claim 16, ~~characterised in that~~ further
comprising:

mixing hot bleed air from the engine with the warm waste air, ~~which~~ that originates from
the cooling of the aircraft's electronic equipment, the mixing of the hot bleed air and the warm
waste air occurring before conveyance ~~is mixed with hot engine bleed air before it is conveyed to~~
the first plurality of hollow chambers.

Application No. 10/582,700
Reply To Office Action Dated April 16, 2008
Response Dated July 16, 2008

22.(Currently Amended) Method in accordance with claim 19,~~characterised in that~~ further comprising:

mixing hot engine bleed air is mixed into from the engine with the warm waste air ~~which~~
that originates from the cooling of the aircraft's electronic equipment to create a first mixture, the
mixing occurring upstream of the first plurality of the hollow chambers of the floor, and ~~that also~~
mixing in additional hot engine bleed air from the engine is mixed into the waste air after the latter
~~has flowed through~~ downstream of the first plurality of hollow chambers of the floor, but
upstream of ~~before flowing through~~ the cargo hold door.

23.(Previously Presented) Method in accordance with claim 16,

characterised in that the panels forming the floor are additionally heated by electricity.

24.(Previously Presented) Method in accordance with claim 16,

characterised in that a forced flow is generated in the hollow chambers.